Aircraft Spruce & Specialty Co.

LASAR® SYSTEM INFO REQUIREMENT SHEET

THE FOLLOWING ENGINE AND AIRCRAFT INFO IS NEEDED TO CORRECTLY DETERMINE LASAR® SYSTEM PART NUMBERS. WITH THIS INFORMATION IN HAND, PLEASE REFER TO L-1502 APPLICATION GUIDE OR TO THE APPROPRIATE APPROVED PARTS MATRIX TO DETERMINE THE CORRECT LASAR® SYSTEM COMPONENT PART NUMBERS.

AIRFRAME AND ENGINE MAKE AND MODEL LASAR® INSTALLATION THROUGH BOTH AIRFRAME AND ENGINE SUPPLE-MENTAL TYPE CERTIFICATES (STC) ISSUED BY THE FAA. FOR THE TEXTRON LYCOMING 320 AND 360 SERIES ENGINES, THE AIRCRAFTS ENGINE MAKE AND MODEL NUMBER MUST BE LISTED IN THE STC NUMBER SA563CH & SE524CH, RESPECTIVELY. INSTALLATION IN EXPERIMENTAL AIRCRAFT DOES NOT REQUIRE REFERENCE TO THE STC.
AIRFRAME MODEL
ENGINE MODEL
ENGINE IGNITION BASE TIMING AND HORSEPOWER RATING THE BASE TIMING ANGLE REPRESENTS THE NUMBER OF DEGREES BEFORE TOP DEAD CENTER THAT THE SPARK PLUGS MUST FIRE, AND IS USUALLY PRINTED ON THE ENGINES DATA PLATE. THIS ANGLE IS IMPORTANT BE- CAUSE IT DETERMINES THE FIRING ANGLE OF THE LASAR® SYSTEM WHEN OPERATING THE BACK UP MODE OP- ERATION. THE HORSE POWER RATING, IN PART, DETERMINES THE PROPER ENGINE PERSONALITY MAP AND IS NEEDED TO IDENTIFY THE CONTROLLER PART NUMBER.
BASE TIMING ANGLE
RATED HORSE POWER
MAGNETO DRIVE TYPE THE DRIVE SYSTEMS OF THE CONVENTIONAL MAGNETOS BEING REMOVED MUST BE KNOWN TO ENSURE THAT THE LASAR® MAGNETOS WILL PROPERLY INTERFACE WITH THE EXISTING ENGINE ACCESSORY DRIVE GEAR AND MOUNT- ING PAD CONFIGURATION. LEFT MAGNETO: DIRECT DRIVE IMPULSE COUPLED IMPULSE COUPLED
SOURCE OF TACHOMETER SIGNAL CURRENTLY USED LASAR® SYSTEMS USE FOUR POLE ROTORS AND DO NOT HAVE TRADITIONAL "P" LEADS OR TACHOMETER BREAK- FRS. THEREFORE COMPATIBILITY OF LASAR® SYSTEMS WITH ELECTRONIC TACHOMETERS WHICH DERIVE AN

INPUT SIGNAL FROM ANY OF THE FOLLOWING SOURCES MUST BE VERIFIED BY THE TACH MANUFACTURER. REFER TO THE LATEST REVISION OF L-1502 INSTALLATION OPERATION AND TROUBLESHOOTING FOR DETAILS OF THE

P LEAD

TACHOMETER SIGNAL GENERATED BY THE LASAR® CONTROLLER.

TACH

HALL EFFECT

AIRFRAME VOLTAGE TWO DIFFERENT LASAR® CONTROLLER DESIGNS ARE AVAILABLE TO WORK WITH EITHER 12 VOLT OR 24 VOLT NOMINAL AIRFRAME BUSSES. LASAR® CONTROLLERS INTERPRET POWER INPUTS BELOW 6 VOLTS AS AN AIR- CRAFT ELECTRICAL FAILURE AND AUTOMATICALLY SWITCH TO BACK UP MODE OPERATION. STARTER MOTOR CURRENT DRAW, BATTERY CONDITION AND LOCATION, AND BATTERY CABLE LENGTH AND CONDITION CAN LEAD TO EXCESSIVE VOLTAGE DROP DURING ENGINE CRANKING AND CAUSE HARD ENGINE STARTING.
12 VOLT AIRFRAME BUS 24 VOLT AIRFRAME BUS
DISTANCE FROM THE CONTROLLER TO BOTH LEFT AND RIGHT MAGNETOS LOW VOLTAGE CONTROL HARNESSES INCLUDE TWO WIRE BUNDLES, EACH TERMINATING IN CONNECTERS. THES ATTACH IN TEN INCH PIGTAIL HARNESSES WHICH ARE PART OF EACH LASAR® MAGNETO. THE DISTANCE FRO THE CONTROLLER TO EACH MAGNETO MUST BE KNOWN TO DETERMINE THE REQUIRED LENGTH OF THE TWO TER- MINATED WIRE BUNDLES. ALL HARNESSES WITH A BASE PART NUMBER LH1004-XXINCLUDE WIRES FOR A CHT TACHOMETER AND COCKPIT ENUNCIATOR LIGHT INTERFACES.
INCHES FROM CONTROLLER TO LEFT MAG
INCHES FROM CONTROLLER TO RIGHT MAG
TYPE OF CYLINDER HEAD TEMPERATURE PROBE CURRENTLY INSTALLED MOST LASAR® INSTALLATION (WITH THE EXCEPTION OF CERTAIN CESSNA 172 AND NON CERTIFIED AIRCRAFT INCLUDE THE CYLINDER HEAD TEMPERATURE CONTROL FEATURE WHICH REQUIRES INPUT FROM A THERMISTOR TYPE TEMPERATURE PROBE INSTALLED IN THE CYLINDER GENERATING THE MOST HEAT DURING THE MAXIMUM ANGLE OF ATTACK CLIMB. THE TYPE OF PROBE CURRENTLY INSTALLED (IF ANY) MUST BE KNOWN TO DETER MINE THE CORRECT REPLACEMENT LASAR® PROBE.
DUAL POINT PROBES ARE AVAILABLE WHICH WILL DRIVE EXISTING CHT INSTRUMENTS AND PROVIDE THE NEEDED CHT SIGNAL TO THE LASAR® CONTROLLER. "J" TYPE PROBES CAN BE IDENTIFIED BY EITHER RED, WHITE, OR YELLOW AND BLACK LEADS, WHILE "K" PROBES USE RED AND BLACK LEADS.
□ NONE □ "J" TYPE □ "K" TYPE
Mag Series numbers (10- NUMBER IF BENDIX)?

What Spark plug do you have?

.....please return to the sales agent who sent this form to you.....

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